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REPORT NO. 89-R-05 AFPEA PROJECT NO. 86-P-139



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QUALIFICATION TESTING OF THE CNU-471/E INFRARED DETECTION SET CONTAINER

HQ AFLC/DSTZ
AIR FORCE PACKAGING EVALUATION ACTIVITY
Wright-Patterson AFB OH 45433-5999

June 1989 `

MOTICE

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ABSTRACT

Aeronautical Systems Division, ASD/VXAL, requested assistance from the Air Force Packaging Evaluation Activity (AFPEA) to design, prototype, and qualify a container for the infrared detection set (IDS) used on Combat Talon II aircraft.

The CNU-471/E prototype container was designed, fabricated and tested at the AFPEA, HQ AFLC/DSTZ, Wright-Patterson AFB, OH 45433-5999. The container is environmentally sealed and outfitted with a humidity indicator, desiccant port, and pressure relief valve. container's item receptacle was designed to interface with the Texas Instrument handling ring for ease in maneuverability. The container is designed to protect one IDS during worldwide shipment, storage, and handling.

The container test plan was derived from Military Specification

MIL-C-5584D. The tests were cond Test Method Standard No. 101, Mil Standard 1489, and Military Speci	lucted in accordance with Federal litary Standard 648, Military RR P)/
Results of the tests conducted on acceptable.	the prototype container are	
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Ted Minds The Mi	Charlie P. Edmonson Chief, AF Packaging Evaluation Activity	

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TABLE OF CONTENTS

	<u>Page</u>
ABSTRACT	. i
TABLE OF CONTENTS	. ii
INTRODUCTION	
BACKGROUND	. 1
PURPOSE	. 1
DESCRIPTION OF TEST CONTAINER	. 1
TEST OUTLINE AND TEST EQUIPMENT	. 1
TEST PROCEDURES AND RESULTS	. 2
TEST NO. 1, WEIGHT TEST	. 2
TEST NO. 2, LEAK TEST	. 2
TEST NO. 3A, CORNERWISE-DROP (ROTATIONAL) (+140°F) TEST	. 2
TEST NO. 3B, EDGEWISE-DROP (ROTATIONAL) (+140°F) TEST.	. 2
TEST NO. 3C, PENDULUM-IMPACT (+140°F) TEST	. 3
TEST NO. 4, LEAK TEST	. 3
TEST NO. 5A, CORNERWISE-DROP (ROTATIONAL) (-20°F) TEST	. 3
TEST NO. 5B, EDGEWISE-DROP (ROTATIONAL) (-20°F) TEST	. 3
TEST NO. 5C, PENDULUM-IMPACT (-20°F) TEST	3
TEST NO. 6, LEAK TEST	. 4
TEST NO. 7, VIBRATION FATIGUE TEST	. 4
TEST NO. 8, LEAK TEST	. 4
TEST NO. 9, REPETITIVE SHOCK (SUPERIMPOSED LOAD) TEST.	. 5
TEST NO. 10, LEAK TEST	. 5
TEST NO. 11, SUPERIMPOSED LOAD TEST	. 5

TEST NO. 12, LEAK TEST	5
TEST NO. 13, STAND OFF TEST	6
TEST NO. 14, MECHANICAL HANDLING TESTS	6
TEST NO. 15, HOISTING STRENGTH TESTS	6
TEST NO. 16, LEAK TEST	7
TEST NO. 17, COVER HANDLE PULL TEST	7
TEST NO. 18, PUNCTURE RESISTANCE TEST	7
TEST NO. 19, LEAK TEST	7
CONCLUSION	8
RECOMMENDATION	8
TABLE 1, CONTAINER TEST PLAN	9
FIGURE 1, CNU-471/E CORNER, SIDE AND LATCH NUMBERING	18
FIGURE 2, CNU-471/E PROTOTYPE CONTAINER	19
FIGURE 3, CNU-471/E ITEM RECEPTACLE AND LIFT RING	19
FIGURE 4, HIGH TEMPERATURE EDGEWISE-DROP	20
FIGURE 5, HIGH TEMPERATURE PENDULUM-IMPACT TEST	20
FIGURE 6, SUPERIMPOSED LOAD TEST	21
DISTRIBUTION LIST	22

APPENDICES:

APPENDIX 1, DETAILED ACCELERATION RESULTS

INTRODUCTION

BACKGROUND: Aeronautical Systems Division (ASD/VXAL), Wright-Patterson AFB OH 45433-5000 requested assistance from the Air Force Packaging Evaluation Activity (AFPEA) to design, prototype, and perform qualification testing on the infrared detection set (IDS) container. The CNU-471/E prototype was designed and fabricated at the AFPEA, HQ AFLC/DSTZ, Wright-Patterson AFB, OH 45433-5999.

<u>PURPOSE</u>: The purpose of this project was to determine if the CNU-471/E container design will protect the contents, one IDS for Combat Talon II aircraft, during worldwide shipment, storage, and handling.

DESCRIPTION OF TEST CONTAINER

The CNU-471/E prototype was subjected to extensive testing. The sides, edges and latches of the container were numbered counterclockwise from the forward end as shown in figure 1.

Design: The CNU-471/E prototype is a controlled-breathing container with a pressure relief valve, humidity indicator, and desiccant port (see figure 2). The container is designed to limit the transmission of shocks to the IDS to 25 Gs when subjected to the conditions in Military Specification MIL-C-5584D. Twelve wide-handle latches (see recomendation) are designed to allow quick access to the container contents without the use of tools. The container's item receptacle was designed to interface with the Texas Instrument handling ring (see figure 3).

Construction: The container consists of aluminum extrusions for the base walls, the lower cover walls, the wall stiffeners, the stacking interface and the skid base. Sheet aluminum is used for the upper cover walls, the top and bottom surfaces and the item receptacle. The item receptacle also uses aluminum bar stock for supports. PPP-C-1752, Type 11, Class 1, (two pound density polyethylene foam) provides cushioning between the item receptacle, the sides, floor and the cover of the container. A silicone gasket provides a seal between the container base and the container cover.

TEST OUTLINE AND TEST EQUIPMENT

Test Plan: Tests were conducted in accordance with AFPEA Test Plan 86-P-139 (see table 1). The tests used were selected to meet the qualification requirements in Military Specification MIL-C-5584D. Test methods and procedures used were as outlined in Federal Test Method Standard No. 101 (FED-STD-101),

Military Standard 648 (MIL-STD-648), Military Standard 1489 (MIL-STD-1489), and Military Specification MIL-C-5584D. Any modifications to the standard procedures are noted in the test plan or the results.

Test Load: All tests were conducted using the IDS test load fabricated at the AFPEA. The test load weighs 92 pounds and simulates the center of gravity and the mass moment of inertia of an IDS as closely as possible.

Test Site: All testing was conducted at the AFPEA, HQ AFLC/DSTZ, Building 70, Area C, Wright-Patterson AFB OH 45433-5999. The equipment required for each test is noted in the test plan.

TEST PROCEDURES AND RESULTS

1. Weight Test

Test Plan No. 1: The container was weighed to determine weight compliance.

Results: The cover weighed 68 pounds, the base weighed 114 pounds, total tare weight of 208 pounds. The results of this test are acceptable.

2. Leak Test

Test Plan No. 2: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2 at 1.00 psig. The vacuum retention test was conducted in accordance with FED-STD-101, Method 5009.2 at 1.00 psig. The failure criteria for the test was 0.025 psig loss during a 30 minute period, after temperature and pressure stabilization.

Results: At the end of the 30 minute test period the pressure loss was 0.007 psig. At the end of the 30 minute test period the vacuum loss was 0.007 psig. The results of this test are acceptable.

3. Rough Handling Tests (+140°F)

a. <u>Test Plan No. 3a</u>: The high temperature cornerwise-drop (rotational) test was conducted in accordance with FED-STD-101, Method 5005.1. Due to the location of the center of gravity the maximum attainable height of the drop was 26 inches.

Results: Visual inspection revealed no external damage to container. A maximum of 10 Gs was obtained during the test.

The Plan No. 3b: The high temperature edgewise-drop (rotational) test (see figure 4) was conducted in accordance with

FED-STD-101, Method 5008.1. Due to the location of the center of gravity the maximum attainable height of the drop was 25 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 12 Gs was obtained during the test.

c. <u>Test Plan No. 3c</u>: The high temperature pendulum-impact test (see figur 5) was conducted in accordance with FED-STD-101, Method 5012. The impact velocity was 7 ft/sec, the height of the drop was 9 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 4 Gs was obtained during the test.

The container was opened after the pendulum-impact test. Visual inspection revealed no damage to the container or the test load. The results of these tests are acceptable. See appendix 1 for detailed acceleration results.

4. Leak Test

Test Plan No. 4: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.013 psig. The result of this test is acceptable.

Rough Handling Tests (-20°F)

a. <u>Test Plan No. 5a</u>: The low temperature cornerwise-drop (rotational) test was conducted in accordance with FED-STD-101, Method 5005.1. Due to the location of the center of gravity the maximum attainable height of the drop was 26 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 22 Gs was obtained during the test.

b. <u>Test Plan No. 5b</u>: The low temperature edgewise-drop (rotational) test was conducted in accordance with FED-STD-101, Method 5008.1. Due to the location of the center of gravity the maximum attainable height of the drop was 25 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 22 Gs was obtained during the test.

c. <u>Test Plan No. 5c</u>: The low temperature pendulum-impact test was conducted in accordance with FED-STD-101, Method 5012. The impact velocity was 7 ft/sec, the height of the drop was 9 inches.

Results: Visual inspection revealed no external damage to the container. A maximum of 20 Gs was obtained during the test.

The container was opened after the pendulum-impact test. Visual inspection revealed no damage to the container or the test load. The results of this test are acceptable.

6. Leak Test

Test Plan No. 6: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.013 psig. At the end of this test, the center latch on each side (total of four) was released to determine if leakage rate would be affected. There was no change in the leakage rate and the results of this test are acceptable.

7. <u>Vibration Fatigue Test</u>

Test Plan No. 7: The vibration fatigue test was conducted in accordance with MIL-STD-648, paragraph 5.3.2. The container was rigidly attached to the platform. A sinusoidal vibration excitation was applied in a vertical direction and cyclically swept for 7.5 minutes at 2 minutes per octave to locate the resonant frequency. Input from 5 to 12.5 Hz was at 0.125 inch double amplitude and input from 12.5 to 50.0 Hz was at 1.0 G. A 30 minute dwell test was conducted at the resonant frequency.

Results: Visual inspection revealed no damage to the container or the test load. A maximum of 4.4 Gs was obtained at the resonant frequency of 9 Hz. The maximum transmissibility obtained was 4.2. A permanent set of 1/8-inch was measured for the bottom cushion and a 3/16-inch for the top cushion. The results of this test are acceptable.

8. Leak Test

Test Plan No. 8: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.005 psig. The result of this test is acceptable.

9. Repetitive Shock (Superimposed Load) Test

Test Plan No. 9: The repetitive shock test was conducted in accordance with MIL-STD-648, paragraph 5.2.2.1 and FED-STD-101, Method 5019.1. 300 pounds of steel plate was banded to the top of the container to simulate a stacked load. The container was not attached to the platform, but restraining blocks were attached to the platform to prevent the container from moving off the platform. The platform was vibrated at 3 to 5 Hz until the container raised from the platform (1/16 inch feeler gauge clearance between the container bottom and the platform), or a maximum platform acceleration of 1.0 G +/- .1 G was achieved. The test was run at the determined frequency for a period of two hours.

Results: Visual inspection revealed no damage to the container or the test load. A maximum of 2 Gs was obtained during the test. A permanent set of a 3/16-inch was measured for the bottom cushion and a 1/4-inch for the top cushion. The results of this test are acceptable.

10. Leak Test

Test Plan No. 10: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.005 psig. The result of this test is acceptable.

11. <u>Superimposed Load Test</u>

Test Plan No. 11: The superimposed load test (see figure 6) was conducted in accordance with FED-STD-101, Method 5016.1. A load of 2429 pounds was placed on top of the container through beams placed to simulate actual loading on the bottom container of a stack of four with a factor of safety of two.

Results: Visual inspection revealed no damage to the container. The results of this test are acceptable.

12. Leak Test

Test Plan No. 12: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.005 psig. The result of this test is acceptable.

13. Stand Off Test

Test Plan No. 13: The stand off test was conducted in accordance with paragraph 4.7.5.1 of MIL-C-5584D. The cover was set on a concrete surface with the stand offs in contact with the floor. A load of 136 pounds was then placed on top of the cover for deflection test. With the load removed, the cover was then slid on the stand offs five feet in each of four directions.

Results: Visual inspection revealed no deflection or deformation with the 136 pound load on top. No physical damage resulting in a loss of functional performance was found following the sliding. The results of this test are acceptable.

14. Mechanical Handling Tests

a. Test Plan No. 14a: The forklift handling test was conducted in accordance with FED-STD-101, Method 5011.1, paragraph 6.2. 1x4 inch boards were used since the forklift used has hard rubber tires. Stack one container on another container and verify forklift entry from all four sides.

Results: During the test the container was stable riding on the times. Visual inspection revealed no damage to the container. Forklift entry was possible on all four sides. The results of this test are acceptable.

h. Test Plan No. 14b: The forklift pushing test was conducted in accordance with FED-STD-101, Method 5011.1, paragraph 6.5.

Results: Visual inspection revealed no functional damage to the container. The result of this test is acceptable.

c. <u>Test Plan No. 14c</u>: The forklift towing test was conducted in accordance with FED-STD-101, Method 5011.1, paragraph 6.6.

Results: Visual inspection revealed no functional damage to the container. The result of this test is acceptable.

15. Hoisting Strength Tests

Test Plan No. 15a: The single ring hoisting test was conducted in accordance with MIL-STD-648, paragraph 5.8.5. The maded container was lifted by a lift ring and suspended for five languages.

Me wits: Visual inspection of the container revealed no damage or deformation. The result of this test is acceptable.

b. <u>Test Plan No. 15b</u>: The 4 ring hoisting strength test was conducted in accordance with MIL-STD-648, paragraph 5.8.3. The container was loaded with 1200 additional pounds and hoisted by all four lift points on the container base simultaneously and left hanging for five minutes.

Results: Visual inspection revealed no damage to the container. The results of this test are acceptable.

16. <u>Leak Test</u>

Test Plan No. 16: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: At the end of the 15 minute test period the pressure loss was 0.004 psig. The result of this test is acceptable.

17. Cover Handle Pull Test

Test Plan No. 17: The cover handle pull test was conducted in accordance with paragraph 4.7.4.1 of MIL-C-5584D. One handle was used to lift the 68 pound cover off the ground. A 182 pound weight was placed in the cover to give a total weight of 250 pounds. The cover was maintained in that position for 5 minutes.

<u>Results</u>: Visual inspection revealed no deflection or permanent deformation to the cover handle or the container cover. The results of this test are acceptable.

18. Puncture Resistance Test

Test Plan No. 18: The pendulum puncture test was conducted in accordance with MIL-STD-1489A, Method 505. The test apparatus used was a simulated forklift tine weighing 70 pounds suspended by wire cables. The tine was pulled straight back until it reached a height of 28 inches above its equilibrium height and released. The tine impact was made to each side and end of the unrestrained container at one inch above the forklift pocket.

Results: Visual inspection revealed no functional damage to the container. The results of this test are acceptable.

19. Leak Test

Test Plan No. 19: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 1.00 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period.

Results: A failure occurred due to an internal crack at a skid weld and a slight leak due to some roughness detected on the cover flange interface contacting the gasket near latch number twelve. Repairs were made by rewelding the skid weld crack and smoothing the rough interface. Retesting of the container for leakage showed no loss of pressure. At end of the retest, the center latches on each side (total of four) were released and no pressure loss was detected. The results of this test are acceptable.

CONCLUSION

The CNU-471/E prototype container provided adequate protection for the contents when tested in accordance with the container test plan.

RECOMMENDATION

It is recommended that one latch from each side (total of four) be deleted from the design as they appear unnecessary to maintain the container sealing integrity during worldwide shipment and storage of one IDS.

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AFPEA PROJECT NUMBER AIR FORCE PACKAGING EVALUATION ACTIVITY (Container Test Plan) 86-P-139 WEIGHT (LBS) CUBE (CU. FT.) QUANTITY CONTAINER SIZE (L x W x D)(INCHES) DATE INTERIOR: EXTERIOR: **GROSS:** ITEM: 22 Jun 89 \$3 x 31 x 38.3 300 92 22.5 1

ITEM NAME MANUFACTURER

Combat Talon II Infrared Detection Set Texas Instruments

CONTAINER NAME CONTAINER COST

CNU-471/E

PACK DESCRIPTION

Aluminum Container

CONDITIONING

As noted below.

	REF STD/SPEC		T	T
TEST NO.	AND TEST METHOD OR PROCEDURE NO'S	TEST TITLE AND PARAMETERS	CONTAINER ORIENTATION	INSTRU- MENTATION
c.	FED-STD-101 Method 5012 [4.7.8]	Pendulum-impact test. Condition at +165°F. Temperature of shock mitigation system at time of test shall be +140°F (+10/-0°F). Impact velocity 7 ft/sec, drop height 9 inches.(4)	One impact on one side and one end, total of two impacts.(3)	Tri-axial accelero- meters, Thermo- couples
4.	LEAK TEST FED-STD-101 Method 5009.2 [4.7.2]	Pneumatic pressure with 1.00 PSI. Test duration not less than 15 minutes with .0125 PSI leakage allowed.	Ambient	Water manometer
5. a.		Cornerwise-drop (rota- tional) test. Condition at -20°F for not less than 24 hours. Drop height 32 inches.(4)	F) One drop on diagonal bottom cor- ners, total of two drops.(5)	Tri-axial accelero- meters
ъ.	FED-STD-101 Wethod 5008.1 [4.7.8]	Edgewise-drop (rota- tional) test. Condition at -20 ^O F for not less than 24 hours. Drop height 32 inches.(4)	One drop on two bottom edges, total of two drops.	Tri-axial accelero- meters

the description of the same opposite those impacted in Test No. 3a.

These diges are opposite those impacted in Test No. 3b or 3c.

APPROVED BY:

Teroline Buckey, Mechanical Engineer Ted Hinds, Chief, Design Br., AFPEA

AFA-D FORM 4 10 PAGE 2 OF

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AFALD FORM, 4

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	Method 50		1 G or 1, between						meters
	[4.7.7.3]		(whicheve						
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	3 70 - 3 7D-1	,1	Pneumatio	pressu	re with	Ambi	ent		Water
	. "Mod 500	19.2	1.00 PSI.	Test	duration	n			manometer
	7.44	,	not less			s			
			with .012 allowed.	.J POI 1	еакаде				
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REPAH	(로 리 영Y:		the Atomic Control By a		APPROVE	D BY:			andri affirm anno merceno menten de la como

AFALD NOV 8, 4

	AIR FORCE PACE			ION	I AC	TIV	ITY	AFPEA PROJECT NUMBER		
		Container T						86-P-3		
CC	ONTAINER SIZE (L x W x I INTERIOR: EX) (INCHES) TERIOR:	WEIGHT GROSS:		S) EM:	CUBE	E (CU. FT.)	QUANTITY	DATE	
		1 x 38.3			92	2.	2.5	1	22 Jun 89	
TEM N	IAME	'	•	1	MANUF	ACTU	RER	•	•	
	bat Talon II Inf	rared Det	ection S	Set			Texa	as Instr	uments	
CONTA	INER NAME						C	ONTAINER CO	DST	
	-471/E									
	DESCRIPTION									
	minum Container									
	TIONING									
	noted below.	Т								
TEST NO.	REF STD/SPEC AND TEST METHOD OR PROCEDURE NO'S	TEST TI	TEST TITLE AND PARAMETERS					ITAINER NTATION	INSTRU- MENTATION	
	CUDEDIMPOSED IO					1				
11.	SUPERIMPOSED LOS	Stack 4	contain	ers	or 1	16	Bottom		Record	
	Method 5016.1	ft high,	whichev	er	is		contai		changes,	
	[4.7.6.1]	greater.					contair		i.e.	
	1	load on times a					under 1	test.	buckling, deforma-	
	1	of 2, le							tions.	
	1	60 minut								
		temperat	ure.							
12.	LEAK_TEST	,				1				
	FED-STD-101	Pneumati				- 1	Ambient	5	Water	
	Method 5009.2	1.00 PSI							manometer	
	[4.7.2]	not less				es				
		allowed.	23 101 1	· Can	uge	ŀ				
12	CMAND-OFF MECH					Í				
13.	STAND-OFF TEST [4.7.5.1]	Place lo	ad two t	ime	s th	ne l	Place o	con-	Visual	
	[40,0302]	cover we					tainer		inspection	
		The cove						oncrete	_	
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	202 AV.			A 804	ROVE					
PEPA	RED BY:	nanical E		1					Br., AFPEA	

	AIR FORCE PAC	CAGING E		ION AC	TIVITY	}	AFPEA PROJECT NUMBER 86-P-139		
CC	ONTAINER SIZE (L x W x INTERIOR: EX	D) (INCHES) TERIOR:	WEIGHT	(LBS)	CUBE (CU.	FT.) QUAN	MTY	DATE	
		1 x 38.3	300	92	22.5	1		22 Jun 8	
TEM N	IAME		I	MANUF	ACTURER				
Com	bat Talon II Inf	rared Det	ection S	Set	T	exas Ins	strume	ents	
CONTA	INER NAME					CONTAINE	R COST		
	-471/E								
	DESCRIPTION	•							
	minum Container								
	rioning								
As	noted below.	1			-1		,		
NO.		TEST TI	TLE AND PAI	RAMETERS		CONTAINER RIENTATION	1	INSTRU- MENTATION	
	NO. AND TEST METHOD OR TEST TITLE AND PAR				n ft at si)	ent			
	. 470			1					
DEDA.	PED BY:			APPROVE	D ST!				

AFALD NOVY 4

	AIR FORCE				ION A	YTIVITY	AFP	EA PRO	JECT NUMBER
			Container T	est Plan)			, -	6-P-1	
CC	ONTAINER SIZE (INTERIOR:	EX	D)(INCHES) Terior: 1 x 38.3	WEIGHT GROSS: 300	(LBS) ITEM: 92	CUBE (CU. 22.5	FT.) QUA	NTITY	DATE 22 Jun 8
TEM N)	I X 30.3	300		FACTURER			22 041. 0.
Com	bat Talon :	II Inf	rared Det	ection S	et	T	exas I	nstru	ments
	INER NAME						CONTAIL	NER CO	BT
CNU	-471/E						Ī		
ACK	DESCRIPTION						_ 		
Alu	minum Conta	ainer							
CONDI	TIONING							· · · · · · · · · · · · · · · · · · ·	
As	noted below	۸.			·				
TEST NO.	REF STD/S AND TEST ME PROCEDURE	THOD OR	TEST TI	TLE AND PAF	RAMETERS		CONTAINE		INSTRU- MENTATION
b.	FED-STD-10 Method 500 Paragraph [4.7.5]	5011.1 The forklift times ph 6.5 should extend under but				and cont	one sone one endainer.		Visual inspection
c.	FED-STD-10 Method 500 Paragraph [4.7.5]	11.1	Forklift towing test. Fr				ends dainer	of	Visual inspection
5.	HOISTING S	TRENG'	TH TEST						
a.	MIL-STD-64 Para 5.8.5 [4.7.4]		Single r test. H from each turn and five min shall be permanen	Ambi	ent		Visual inspection		
b.	MIL-STD-64 Para 5.8.3 [4.7.4]		Four ring hoisting test. Hoist container loaded to five times the gross weight of a single container by all lift points simultaneously and leave hanging for 5 minutes. There shall be			5	ent		Visual inspection
COMMI	ENTS:		1						
REPA	RED BY:				APPROVE	D SY:			
			nanical E		I -				Br., AFPEA

X D) (INCHES) WEIGHT (LBS) CUBE (CU. FT.) QUANTITY DATE EXTERIOR: GROSS: ITEM:	AIR FORCE PACKAGING EVALUATION ACTIVITY (Container Test Plan) AFPEA PROJ 86~P-1:									
		ITEM:	GROSS:	TERIOR:	EX	RIOR:		C		
x 31 x 38.3 300 92 22.5 1 22 Jun 8			300	1 x 38.3	3 x 3			TEM I		
MANUFACTURER The section Col				manad Dab	T T.		••			
Infrared Detection Set Texas Instruments CONTAINER COST		- 1	eccion s	rared bed	r Tur	NAME				
JONTAINEN JOST	John					1/E	-471	CNU		
		··	****			RIPTION		_		
er					iner	um Conta	minu	Alu		
	· · · · · · · · · · · · · · · · · · ·					NG	TIONIN	CONDI		
					•	ed below	note	As		
		METERS	LE AND PAF	OD OR	REF STD/SP D TEST MET PROCEDURE	AND	TEST NO.			
no damage or permanent deformation. (Legs of the sling should be 30° from the horizontal).		gs of be 30 ⁰	ion. (I g should	deformat the slin			1			
Pneumatic pressure with 1.00 PSI. Test duration not less than 15 minutes with .0125 PSI leakage allowed. Water manometer	Ambient	luratio minute	. Test than 15	1.00 PSI not less with .01		AK TEST D-STD-10 thod 500	FED	16.		
PULL TEST Apply a force of 250 lbs on a cover handle in four directions that service loads are possible. There shall be no damage or permanent deformation. Ambient Scale	Ambient	a force of 250 lbs over handle in irections that e loads are le. There shall damage or			E PUI	VER HAND .7.4.1]		17.		
Impact will be made at a point 1/2 inch above the enclosed forklift pocket of the container base. There shall not be any damage affecting container base, total of two impacts. Visual inspection one end of the container base, total of two impacts.	one side one end contained base, to	ANCE TEST Impact will be made at a point 1/2 inch above the enclosed forklift pocket of the container base. There shall not be any damage affecting con-			9A	CTURE R L-STD-14 thod 505	Met	ં ઇ •		
of the container base. There shall not be any damage affecting con- two impacts.	base, to	base. e any con-	o <mark>ntaine</mark> r all not ffecting	of the contract of the contrac		- -		ર્પા અન્ય દી હ		

AFALD NOT 4

	AIR FORCE				ION AC	TIVITY	AFPEA	EA PROJECT NUMBER	
' <u></u>		((Container T	est Plan)				P-139	
CC	NTAINER SIZE			WEIGHT		CUBE (CU.	FT.) QUANT	ITY	DATE
	INTERIOR:		TERIOR: 1 x 38.3	GROSS: 300	I TEM: 92	22.5	1		22 Jun 89
ITEM N		33 A 3.	L X 30.3	300		FACTURER			22 Juli 59
	oat Talon	TT TNE	Dot	estion C			xas Inst		. 4
	INER NAME	TT TUT	rared Det	ection 5	ec	16			its
							CONTAINER	R COST	
	-471/E DESCRIPTION		· · · · · · · · · · · · · · · · · · ·				<u> </u>		
		•							
	minum Cont	ainer							
	FIONING								
As 1	noted belo								
TEST NO.	REF STD/S AND TEST ME PROCEDURE	THOD OR	TEST TI	TLE AND PAR	AMETERS		CONTAINER RIENTATION		INSTRU- MENTATION
19.	LEAK TEST FED-STD-1		Danimati	c pressu	so with	n Ambie		Wa	ter
	Method 50		1.00 PSI		re with duratio		ent	''	nometer
	[4.7.2]	03.2		than 15				1	
			with .01	25 PSI 1					
			allowed.						
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COMME	NTS:								
PREPAR	ED BY:				APPROVE	BY:			· · · · · · · · · · · · · · · · · · ·
Carol	ine Buckey	, Mech	anical E	ngineer	Ted Hi	nds, Chi	ef, Desi	gn Br	., AFPEA

AFALD SET 4

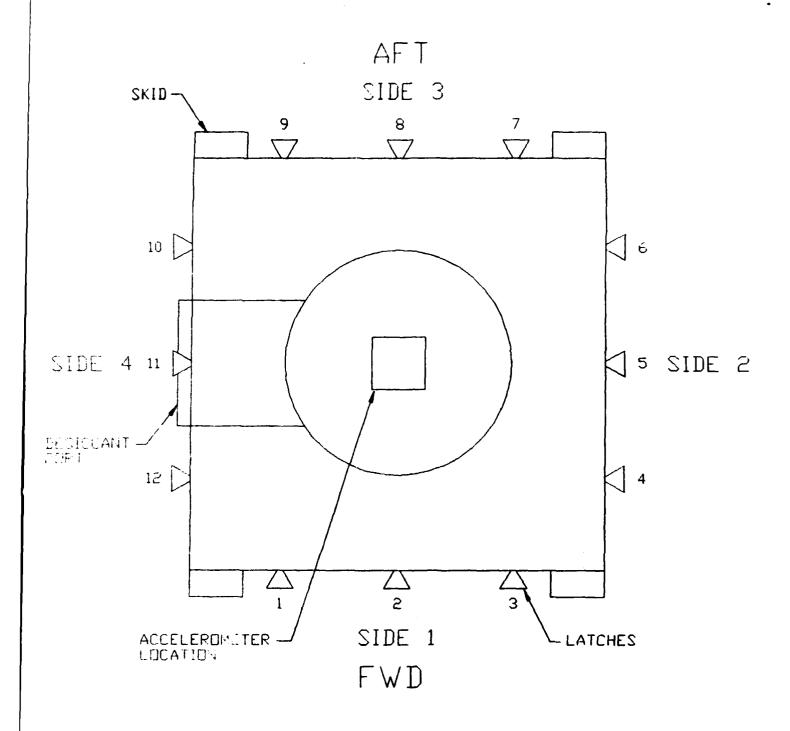


FIGURE 1 - CNU-471/E Corner, Side and Latch Numbering

Figure 2 CNU-471/E Prototype Container.



Figure 3

CNU-471/E

Item

Receptacle
and Lift Ring.



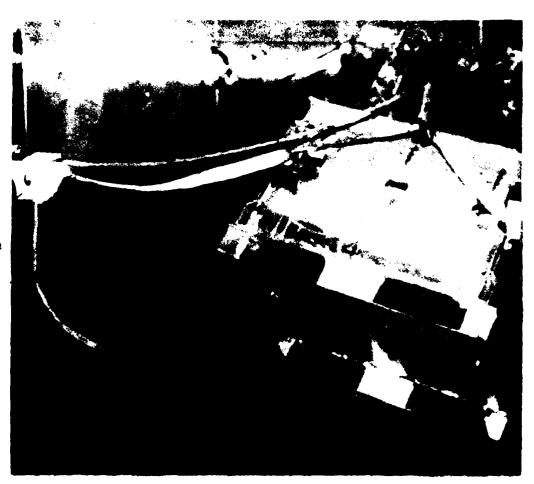


Figure 4

CNU-471/E High Temperature Edgewise-Drop.



CNU-471/E High Temperature Pendulum-Impact Test.

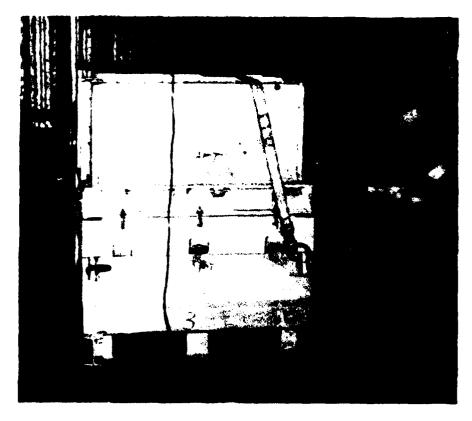
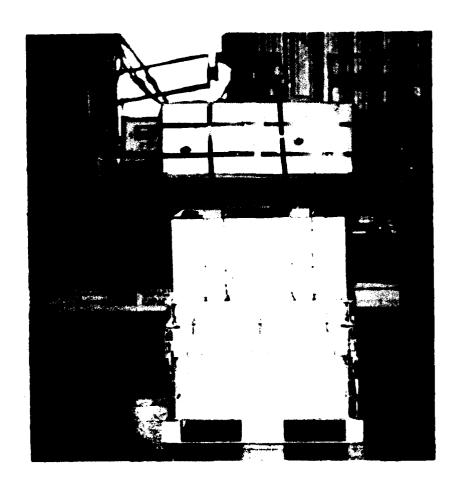


Figure 6

CNU-471/E
Superimposed
Load Test.



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OO-ALC/DSTD Hill AFB UT 84406	1
SA-ALC/DSTD Kelly AFB TX 78241	1
SM-ALC/DSTD McClellan AFB CA 95652	1
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NO MA Deckering Change and	1
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	_
DLSIE/AMXMC-D	1
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Ft Lee VA 23801-6034	
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ASD/VXAL	2
Wright-Patterson AFB OH 45433	
ASD/VXA	1
Wright-Patterson AFB OH 45433	

CNU-471/E CONTAINER - DETAILED ACCELERATION RESULTS

HIGH TEMPERATURE ROUGH HANDLING TESTS (+140°F)

Impact	Position	Accelerometer readings Resultant	(Gs)
26" rotational drop 25" rotational drop 25" rotational drop 24" rotational drop 7 ft/sec pendulum-impact 7 ft/sec pendulum-impact	Corner 1-4 Corner 2-3 Side 1 Side 4 Side 3 Side 4	10 9 12 10 4 3	

1. No damage to the container or the test load.

LOW TEMPERATURE ROUGH HANDLING TESTS (-20°F)

Impact	Acceler Position	rometer readings (Gs) Resultant
26" rotational drop 26" rotational drop 24" rotational drop 25" rotational drop 7 ft/sec pendulum-impact 7 ft/sec pendulum-impact	Corner 3-4 Corner 1-2 Side 2 Side 3 Side 1 Side 2	22 18 18 22 20 13

1. No damage to the container or the test load.

VIBRATION FATIGUE TEST

Natural frequency 9.0 Hz

(input: 1.04 G peak, 0.125 inch double amplitude)

Resultant

Maximum	Acceleration (Gs,	peak	to	peak)	4.4
Maximum	Transmissibility	_		- '	4.2

1. No damage to the container or the test load.